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## 2019 William "Bill" Tynan Memorial AMSAT Space Symposium



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# The New AMSAT® CubeSat Simulator: A New Tool for Education and Outreach

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**Abstract**—The AMSAT CubeSat Simulator is a new tool for education and demonstrations. It can be used in a classroom or training setting to introduce the basics of nanosatellites, or it can be used to teach STEM (Science Technology Engineering and Math) concepts, or it can be a stepping stone in a project to build and launch an actual CubeSat into space. This paper describes a proof-of-concept prototype of the AMSAT CubeSat Simulator built for the 2018 AMSAT Space Symposium, which provides similar functionality to the original ARRL ETP CubeSat Simulator built by Mark Spencer, WA8SME, in 2009. We hope to get useful feedback and suggestions from this paper which will enable us to move ahead with the project.

**Keywords**—cubesat, simulator, telemetry, AMSAT, ham radio, raspberry pi

## I. INTRODUCTION

The AMSAT CubeSat Simulator is a new tool for education and demonstrations. It can be used in a classroom or training setting to introduce the basics of satellites, or it can be used to teach STEM (Science Technology Engineering and Math) concepts, or it can be a stepping stone in a project to build and launch an actual flight model CubeSat.

We have observed that too many CubeSats have been “DOA” (Dead On Arrival) upon deployment on orbit, which is a shame given the time, effort, and expense that was invested in them. An important use of the CubeSat Simulator is to help new CubeSat builders to learn to “crawl before they walk or run.” With the assistance of AMSAT, a new group embarking on a CubeSat mission would not start with a “real” flight model (FM) structure, but instead learn or “practice” first by producing either an engineering test unit (ETU) or, before that, with this CubeSat Simulator.

There are many benefits to building a simulator including groups who wish to have a simple, operating, “flight-like” model for education purposes. We believe that educators and AMSAT Ambassadors are our main audience, but others will take notice as well.

This paper describes a proof of concept prototype of the AMSAT CubeSat Simulator which we will refer to as the CubeSat Sim. We hope to get useful feedback and suggestions from this paper and presentation, which will enable us to move ahead with the project. As the current version is a work in progress prototype, a detailed schematic is not provided in this

paper. Instead, high level design considerations and architecture are described. Once it is finalized, a future article in the AMSAT Journal will provide complete schematics and details.

## II. BACKGROUND ON THE ARRL ETP CUBESAT SIMULATOR

Mark Spencer, WA8SME, developed a CubeSat Simulator about ten years ago, which is the foundation and inspiration for this simulator. The AMSAT Journal published his substantial results in two major articles which are available online [1], [19].

One of the most notable features of the ARRL ETP CubeSat Simulator was its many lessons on Attitude Determination and Control (ADAC), which is a critical subsystem of all satellites big and small, not just all CubeSats. Mark placed a solar cell on each of the six sides of the cube structure with the primary function to charge the “flight” batteries. He used spreadsheet plots of the output voltage measurements of the cells to determine the spatial orientation and rotation rate of the CubeSat to the source of illumination, and thus its attitude with respect to the earth and an observer on the planet’s surface. This served as a crude, but correct, educational method for his series of inquiries and challenges for students to investigate.

Mark not only successfully designed and built the hardware, but he developed a wise philosophy on how to effectively present the material and hardware to teachers, students, fellow amateur radio operators and community groups. His goal was (and still is) to promote and facilitate the advancement of space technology literacy. In doing so, Mark cautions that in a school classroom, we must allow the teacher to do the teaching, and to make the connections to the education standards and benchmarks. Thus, AMSAT and the hams ought to only provide instruction to these teachers and the hardware, but should be ready to advise and assist.

## III. GOALS AND REQUIREMENTS FOR THE AMSAT CUBESAT SIMULATOR

The main goal of the project is to replicate as much as possible the functionality of the original ETP CubeSat Simulator [1]. We have tried to use modern materials and utilize software and general-purpose hardware as much as possible, leveraging Software Defined Radio (SDR) concepts. We have tried to balance cost with performance, usually compromising on the performance to keep the cost of the CubeSat Simulator low. Some other goals are listed here:

- Make all hardware and software freely available through open source.
- Have a robust design that makes maintenance simple.
- Have a look and feel that is as close to a CubeSat engineering model as possible.
- Have functionality that emulates a real CubeSat.
- Have a modular design that allows different subsections to be swapped out for alternatives.
- Build a community to support and extend the design.

Our initial goal was to have a proof-of-concept prototype to show off at the 2018 AMSAT Space Symposium. We believe that with the proof of concept described in this paper we have met most of these goals. Writing this paper and presenting at the Space Symposium are the first steps towards building a community that is excited to make, modify, and extend this CubeSat Simulator.

#### IV. SIMULATOR GOALS

The subsystems of the CubeSat Simulator are shown in Figure 1 and described in the sections below.

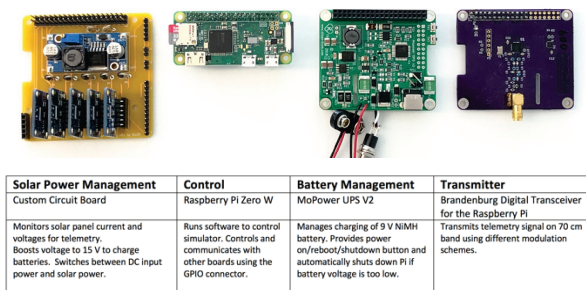


Fig. 1. Proof of concept Prototype Boards and Functions

The Control Subsystem, implemented as a Raspberry Pi Zero W [2], controls all aspects of the CubeSat Sim. It collects data on the operation and status of the CubeSat Simulator, formats the telemetry and transmits it to the Ground Station for display and analysis. The General-purpose Input Output (GPIO) header allows the boards to be stacked and provides access to the various buses. The stacked boards are shown in Figure 2. In this picture, the order of the boards from the bottom is the Solar Power Management board, the Control System board, the Battery Management board, and the Transmitter board. The Battery Management System, implemented using the MoPower V2 UPS (Uninterruptable Power Supply) [3], charges and monitors the rechargeable batteries using power from the Solar Power Management Subsystem, and communicates with the Control Subsystem using the GPIO. Battery voltage, current, and charging status are reported to the Control System. It also sends shutdown and reboot commands to the Control Subsystem based on button presses. The Transmitter Subsystem, implemented as a Digital Transceiver for the Raspberry Pi board, transmits a 70cm telemetry beacon and is configured and controlled by the Control System over the GPIO connector. The

Solar Power Management Subsystem, implemented as a custom circuit board, takes up to six solar panels and combines their outputs and boosts the output to 15 V to charge the batteries. It also switches between solar power and DC input power supply to the Battery Management Subsystem. Voltage and current monitoring is implemented by the Control System using over the GPIO connector. Solar cells provide power to the Solar Power Management System. The Ground Station, implemented as a PC with RTL-SDR USB dongle and free software, receives and decodes the CubeSat Simulator telemetry. The data can be imported into a spreadsheet (such as Excel), the telemetry data can be analyzed and plotted and used to teach satellite or STEM concepts.

Each of these subsystems is described in more detail in the following sections. The interconnection of the subsystems is shown in the block diagram of Figure 3.

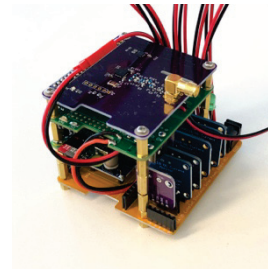


Fig. 2. Proof of concept Prototype Stack

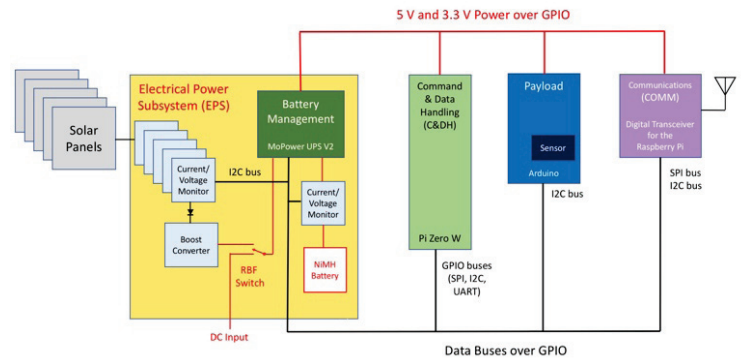


Fig. 3. CubeSat Simulator Subsystem Block Diagram

#### A. Raspberry Pi Zero W

The Raspberry Pi Zero W is a small form factor and low power usage version of the popular Raspberry Pi single board computer, running the Raspbian Stretch version of Debian Linux. It is similar to the common Raspberry Pi 3B or 4B but is about half the width and does not have an ethernet port or full-sized USB ports. Instead, it has a micro USB port and a mini HDMI port. The Pi Zero W does have Wi-Fi and Bluetooth connectivity. We need to disable the Bluetooth functionality as it changes the clock rate of the Raspberry Pi when the UART is enabled, causing communication issues with the Digital Transceiver for the Raspberry Pi board. The Raspberry Pi does not have a hard drive. Instead, storage is a micro SD card of at least 16GB. During the initial install and configuration, a Raspberry Pi unfortunately needs to be connected to a monitor,

keyboard, and mouse. As a result, adapters for the micro USB and mini HDMI are needed. However, once the Wi-Fi is configured and SSH (Secure Shell remote login) is enabled, the Pi can be accessed using a PC (running PuTTY on a PC or ssh on a Mac or Linux) over the network. The Pi Zero W runs on 5 V power which can be provided via the a micro USB cable or via the GPIO.

The Pi Zero W draws about 120 mA, compared to about 400 mA for the Pi 3B+. The Raspberry Pi General-Purpose Input Output (GPIO) bus is used to interface with the other boards in the CubeSat Sim. In order to connect multiple GPIO devices to a Pi, either a GPIO expansion board (such as a GPIO expansion board), or the GPIO headers need to be replaced with stackable headers. Stackable headers provide a better form factor without having to offset each board with the expansion board. The Pi Zero W usually comes without the GPIO header pins installed, making it easy to instead install a stackable header. We also installed a stackable header on the GPIO on the MoPower UPS V2 board.

The GPIO contains the three buses used to communicate with various modules: the UART, the I2C bus, and the SPI bus. The use of each of these buses will be described by the subsystem that uses them to communicate with the Pi.

The software for the Raspberry Pi for this project is open source and available at GitHub [4]. The code is written in C language and has been tested on a Raspberry Pi 3B, 3B+, 4B, and Pi Zero W. Instructions on how to download and compile the software is available there. The software is run as a systemd service to automatically start on boot. The various libraries listed in Table 1 are utilized in the software

TABLE I. TABLE 1. SOFTWARE LIBRARIES USED IN RASPBERRY PI

Library	Use	URL
DigitalTxRXPi	Open source C language library for the Digital Transceiver for the Raspberry Pi board that controls the AX5043 chip, written by Jonathan Brandenburg, KF5IDY	[5]
WiringPi	Open source C library for GPIO, SPI, and I2C bus access on the Raspberry Pi	[6]
MoPower	Python 3 library that communicates between the Raspberry Pi and the MoPower UPS v2 board	[3]

### B. MoPower UPS V2 Board

This board provides Uninterruptable Power Supply (UPS) functionality to a Raspberry Pi by providing battery power via the GPIO interface. This particular board was chosen because of the telemetry functionality built-in. A key function of the board is battery charging and management of the Nickel Metal Hydride (NiMH) batteries. A 15 V AC/DC powerpack is the primary battery charger, with the solar panels, through the Solar Power Management board as secondary. NiMH batteries were chosen to avoid the shipping and packaging requirements needed for Lithium-ion batteries. For classroom and experiment modes, a 200 mAh 9 V NiMH battery will used so that charging and discharging voltage changes can be seen in real-time. Also, this makes it easier to show the automatic shutdown after a period of eclipse. For extended demonstrations and display at events such as hamfests, a bank of six 800 mAh AAA or 2500

mAh AA NiMH batteries can be used to give many hours of operation.

The UPS board has a built-in microcontroller which monitors battery and charging voltages and currents, and communicates with the Raspberry Pi through the GPIO UART (Universal Asynchronous Receiver/Transmitter) using a simple 19.2 kb/s text interface. Note that the Bluetooth interface must be disabled or the clock rate changes when the UART is enabled, making the Digital Transceiver board unreachable. The Raspberry Pi runs Python 3 code that communicates with the board and safely shuts down the Raspberry Pi when the battery voltage drops below 7.4 V. This avoids file corruption on the SD card that is common if the Pi loses power suddenly.

A momentary push button switch is also provided to reboot or startup (press and release) or shutdown (press, hold for three seconds then release) the Pi. The microcontroller can also be programmed for certain events. We could use this functionality to automatically shut down the Pi after an extended period of eclipse, and to start up the Pi from a shutdown state when the RBF pin is removed and the solar panels are illuminated for a period of time. In addition, the MoPower board has four 10-bit analog inputs (0 – 1.1 V) which could be used to gather additional telemetry information from sensors in the future.

### C. Digital Transceiver for the Raspberry Pi Board

The Digital Transceiver for the Raspberry Pi board is a transmitter/receiver for various digital modulation schemes based on the AX5043 chip. The board was designed by Jonathan Brandenburg, KF5IDY, of Brandenburg Tech [7]. The drivers and API library are available on GitHub [5]. The board supports both the I2C (Inter-Integrated Circuit) interface and the SPI (Serial Peripheral Interface) but only the SPI interface is used in the current code.

The board supports a wide range of data rates and modulation schemes. In the initial proof-of-concept prototype, two modes are utilized: CW (Continuous Wave, i.e. Morse code, also known as On-Off Keying or OOK) and 1200 bps AFSK (Audio Frequency Shift Keying) with AX.25 packet formatting. The transmitter is configured to use the 70cm amateur radio band.

### D. Solar Power Management Board

Power to the CubeSat in flight mode is provided by the batteries and the solar panels. For the 1U model, five 65mm by 65 mm 5.5 V 110 mA panels are used, mounted on the +X, -X, +Y, -Y, and +Z sides. For the model with solar panel wings, two 160 mm x 70 mm 7.2 V 200 mA panels are mounted on the wings. Under room illumination, minimal power is delivered by the solar panels; however, a 250 W halogen work lamp placed about 10 cm away produces about half the rated power for indoor usage. Placing the CubeSat Simulator on a small motorized turntable spinning at a few rotations per minute in front of a work lamp can simulate a CubeSat spinning in sunlight on orbit.

On the +Z side (top) of the CubeSat Sim, there are a number of controls and switches and a quarter-wave antenna that connects via a 90-degree adapter and an SMA cable to the Digital Transceiver board. A barrel jack, 5.5 mm x 2.1 mm center positive, is provided for charging the NiMH batteries with

12 - 15 V DC input, such as from an AC power pack. The Remove Before Flight (RBF) pin switch is implemented as a 3.5 mm TRS (Tip Ring Sleeve) plug and jack. To switch the source of power between the DC jack and the solar panels, we use the two switches in the jack that are open when a 3.5 mm plug is plugged in, and closed when the plug is unplugged. When plugged in, the input power to the MoPower UPS is the DC Power Input jack. When unplugged, the input power to the MoPower UPS is the output of the Solar Power Management board. The push button switch is wired to SW3 on the MoPower board, and is used to reboot, shutdown, and startup the CubeSat Sim.

Each solar panel on the CubeSat Simulator is connected together through a 1N5817 Schottky barrier diode rated at 20 V and 1 A to prevent reverse current flow. The 1N5817 was chosen for its low forward voltage drop of about 0.45 V. Each panel's current and voltage is monitored using an Adafruit INA219 board that uses the I2C bus (Inter-Integrated Circuit, pronounced I-squared-C) to communicate with the Raspberry Pi. The Raspberry Pi has three I2C buses, available on the GPIO interface, in which the Pi acts as the bus master. A DC-DC adjustable step-up power converter module [8], based on the ML6009 chip, acts as a boost converter circuit, to increase the 5.5 V from the solar panels to 15 V needed for the MoPower board to charge the NiMH batteries.

### E. Ground Station

The ground station can be any radio receiver capable of receiving FM and CW on the 70 cm band. Since a PC is required to decode the telemetry, a great option is to use a Software Defined Radio (SDR) dongle such as an RTL-SDR [9] (\$20 on Amazon.com) or the FUNcube dongle made by AMSAT-UK [10] and a PC. Any SDR software could be used including the freeware SDR# (pronounced SDR sharp) [11] or HSDR [12], or the open source Gqrx [13]. Figures 4 and 5 show SDR# tuning the CW and AFSK telemetry.

Other software is needed to decode the telemetry information. For the CW telemetry, there are various decoding programs. For the AFSK, the open source Qtmm AFSK 1200 [14] works well, which is a Qt version of the versatile MultiMon software. Note that some kind of speaker-to-microphone audio patching software is needed to pipe the audio output from the SDR software into the microphone input of the decoding software. For Windows, we have used VBCABLE Virtual Audio Device, which is licensed as donationware [15]. The raw telemetry is then copy and pasted into a spreadsheet to decode and display the actual telemetry values. For fun, we chose the AO-7 (AMSAT OSCAR 7 satellite, the oldest functional satellite on orbit, launched in 1974) telemetry format (e.g. hi hi 1aa 1bb 1cc 1dd 2ee ... where aa, bb, cc, and dd are two decimal digits that encode channel 1 data, etc.). Other formats can be added in the future, providing "emulation" for other satellites.

Note that the spreadsheet that we use is a modification of the actual "Morse Code Telemetry Frame Decoder for AMSAT-OSCAR-7" by Jan King, W3GEY, and Jim White, WD0E, as shown in Figure 6. Formulas added to the spreadsheet take pasted telemetry frames and separates (parses) them into channels for display, which are then decoded into actual values.

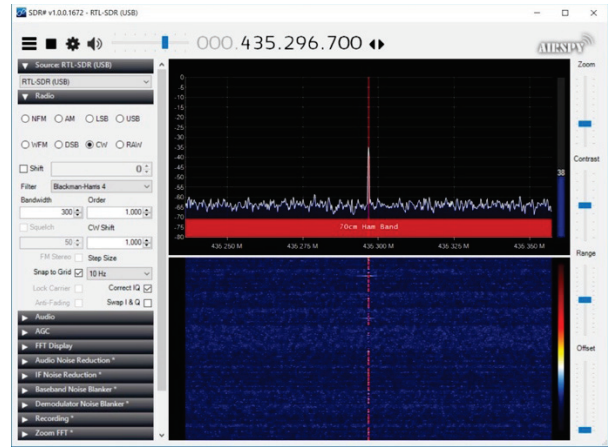


Fig. 4. SDR# Decoding CW Telemetry.

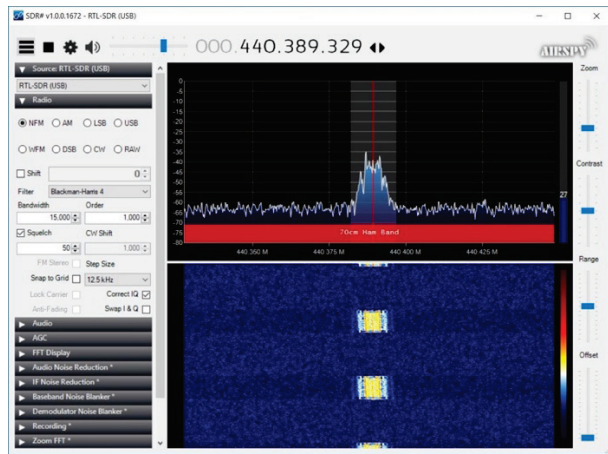


Fig. 5. SDR# Decoding 1200 bps AFSK AX.25 Telemetry.

A	B	C	D	E	F	G	H	I	J	K	L	M	N	O	P	
Morse Code Telemetry Frame Decoder for AMSAT-OSCAR-7																
1	23-Jun-02															
2	W3GEY/WD0E															
3	MC TLM Channel:	Function:	TLM Count:	Scaling Factor:	Original QDR:	Units:	Comments:									
4	1A	Total Array Current	00	0	0ma		May have always been broken. Not sure.									
5	1B	V1 Quad Array I	00	00	400ma		Some bases in this data									
6	1C	V2 Quad Array I	00	10	400ma		Some bases in this data									
7	1D	V3 Quad Array I	00	10	400ma		Some bases in this data									
8	1E	V4 Quad Array I	00	10	400ma		Some bases in this data									
9	1F	V5 Quad Array I	00	10	400ma		Some bases in this data									
10	2A	V1 Quad Array I	00	10	400ma		Some bases in this data									
11	2B	V2 Quad Array I	00	10	400ma		Some bases in this data									
12	2C	V3 Quad Array I	00	10	400ma		Some bases in this data									
13	2D	V4 Quad Array I	00	10	400ma		Some bases in this data									
14	2E	V5 Quad Array I	00	10	400ma		Some bases in this data									
15	3A	1/2 Reg. RF Power Output	75	0.08	1.50mW		Looks on the safe side of normal. Not very above an 432 kHzpt.									
16	3B	1/2 Reg. PA Temp	00	10	0.00C		Not read if channel is broken.									
17	3C	Battery Voltage	00	7.9	6.30V		May be measuring BCR output voltage with no battery load but, with spacecraft load									
18	3D	Battery Charge/Discharge I	00	100	100ma											
19	3E	Battery Charge/Discharge V	00	0.00	0.00V											
20	3F	Battery Temperature	00	21.8	15.30C		Normal for in-use with Regulator #2 Selected									
21	4A	Regulator #1 Voltage	00	21.7	14.40V											
22	4B	Regulator #1 Temp	00	21.8	15.30C											
23	4C	Regulator #2 Voltage	00	21.8	15.30V											
24	4D	Regulator #2 Temp	00	21.8	15.30C											
25	5A	1/2 Reg. PA Temp	00	21.8	15.30C		PA is working, this looks right, guess other S/C temps									
26	5B	1/2 Reg. PA Current	01	11.7	11.60ma		Probably means 'zero' with a 1 count base error.									
27	5C	1/2 Reg. PA Voltage	00	21.8	15.30V		This looks right.									
28	5D	1/2 Reg. PA Power	00	21.8	15.30W		This looks right.									
29	6A	Regulator #1 Power Output	00	0	0ma		Regulator is OFF Normal									
30	6B	Regulator #1 Power Output	00	0	0ma		Regulator is OFF Normal									
31	6C	Regulator #2 Power Output	00	0	0ma		Regulator is OFF Normal									
32	6D	Regulator #2 Power Output	00	0	0ma		Regulator is OFF Normal									
33	7A	Midrange TLM Calibration Value	01	0.11	0.11V		Spec = 0.10V +/- 0.1V									
34																
35	INSTRUCTIONS: The results are given in this column.															
36	Insert the "x" value into this column for the correct "channel" row.															
37	Enter data into the blue cells only.															
38	Write your telemetry down in rows and															
39	columns as shown to the right. Rows															
40	1-4 are the telemetry matrix formatted as shown here,															
41	and the appropriate row (M then 0D).															
42	1A	1B	1C	1D	1E	1F	1G	1H	1I	1J	1K	1L	1M	1N	1O	
43	2A	2B	2C	2D	2E	2F	2G	2H	2I	2J	2K	2L	2M	2N	2O	
44	3A	3B	3C	3D	3E	3F	3G	3H	3I	3J	3K	3L	3M	3N	3O	
45	4A	4B	4C	4D	4E	4F	4G	4H	4I	4J	4K	4L	4M	4N	4O	
46	5A	5B	5C	5D	5E	5F	5G	5H	5I	5J	5K	5L	5M	5N	5O	
47	6A	6B	6C	6D	6E	6F	6G	6H	6I	6J	6K	6L	6M	6N	6O	

Fig. 6. "Morse Code Telemetry Frame Decode Spreadsheet for AMSAT-OSCAR-7" with CubeSat Simulator Data

Most of the telemetry values in the AO-7 format are used by the CubeSat Simulator.

This ground station option requires the installation of SDR dongle drivers on a user's PC, which is a manual process. There are some possible ways to avoid doing this. One would be to use a Raspberry Pi 3B or 3B+ with either a Digital Transceiver for the Raspberry Pi board configured to act as a receiver, or a Pi with an RTL-SDR dongle. This Raspberry Pi could be used

directly with a monitor, keyboard and mouse, or it could be accessed via a PC using a VNC connection over a network, or a direct USB serial connection. Another interesting variant would be to build a ground station with an OpenWrt Linux Wi-Fi router running a `rtl_tcp` server [16] with an RTL-SDR dongle. The `rtl_tcp` server software allows any PC connected to the Wi-Fi network to connect to the RTL-SDR radio using SDR# and receive the IQ (In-phase and Quadrature) data in a stream which they can then demodulate and decode on their own PC. This would also have the advantage of a Wi-Fi network that the CubeSat Simulator would always connect to, making access and configuration easy. Plugging in the Ethernet of the Linux Wi-Fi router to the internet would allow software updates and even remote access to the CubeSat Sim. This same approach that allows users of SDR# to connect to the server and demodulate and decode on their own PC could allow multiple participants in a room to do this, or even for remote participants on a web conference, for example, if the network access was properly configured.

#### F. CubeSat Simulator Frame

This proof of concept prototype has concentrated on functionality, with form taking a lower priority. Also, as electrical engineers, we have stayed in our comfort zone of electronics and software rather than spending a lot of time and energy on the physical design.

We have produced two frames to house the board stack. One is based around 3D printed 1U CubeSat parts from the website Thingiverse [17]. Fully threaded 3 mm steel rods were used as the structure, and clear plexiglass shelves and panels were used for mounting boards and solar panels. **Authors' Note:** The latest frame is fully 3D printed for simplicity. The Raspberry Pi stack was housed in this frame, and is shown in Figure 7 below. Figure 8 shows a frame using the original ETP CubeSat Simulator sheet metal boxes with "wings" for the solar panels.

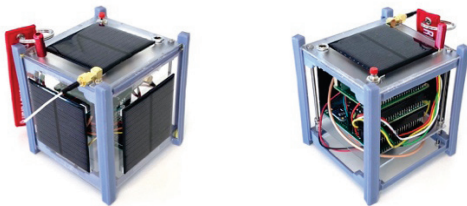


Fig. 7. Figure 7. CubeSat Simulator 1U Frame using 3D Printed Parts.



Fig. 8. Figure 8. CubeSat Simulator Winged Frame using Original ETP CubeSat Simulator Sheet Metal Frame.

#### G. Cost

The approximate cost of making the CubeSat Simulator is around \$300, or \$400 including a ground station, lamp, and turntable. An updated Bill of Materials is available at <http://cubesatsim.org/bom>.

## V. APPLICATIONS AND USES OF THE CUBESAT SIMULATOR

Various educational exercises with a CubeSat Simulator are described in Mark Spencer's paper [19]. This new CubeSat Simulator can be used in many of these exercises. In addition, we anticipate that the CubeSat Simulator will be useful to teams planning to build a CubeSat to gain experience. Also, we hope that emulation for additional satellites will provide practice and training in copying telemetry from new and existing satellites in a classroom setting. We plan to write an article in the AMSAT Journal describing activities and exercises with the CubeSat Simulator.

There is one interesting way in which users without an RTL-SDR radio and antenna can participate in demos using only an internet connected PC. If a remotely accessible SDR server is receiving the CubeSat Simulator transmissions, other users to connect to the server and then do their own decoding of the telemetry. For example, the OpenWebRX [18] server software allows web browser users to connect to the server and demodulate and decode on their own PC could allow multiple participants in a room to do this, or even for remote participants on a web conference, for example, if the network access was properly configured.

Currently, the CubeSat Simulator supports two telemetry modulation schemes: CW and 1200 bps AFSK AX.25. In both cases, the telemetry is encoded in AO-7 format. In the future, support for additional telemetry modulation schemes and telemetry formats can be added. One interesting thing to do in the future would be to implement the Data Under Voice (DUV) 200 bps FSK telemetry used by the AMSAT Fox satellites. This "Fox emulation mode" would allow the use of the FoxTelem software to decode the telemetry. Also, in the future, developers of a new CubeSat could contribute an emulation mode for an upcoming satellite. This would allow amateurs to practice decoding telemetry and using the telemetry analysis tools before launch, avoiding the scramble after a CubeSat is launched and potential loss of initial telemetry data. At AMSAT, we can also ensure that we provide a "GOLF emulation mode" as our next generation of satellites are developed, providing excitement and training on telemetry reception prior to launch.

## VI. NEXT STEPS

Our next steps are to get community feedback on the design and implementation. We are particularly looking for improvements for the frame, to make it sturdy and easy to build.

We plan to fully document hardware and software online and in the AMSAT Journal so that others can build their own. We are also looking for Beta testers to test it out in public demonstrations and classroom exercises and provide feedback. Please contact us directly if you are interested. Teachers are encouraged to write and share example Lesson Plans at specific age and grade-appropriate levels to help educators incorporate the simulator in classes and extracurricular activities.

AMSAT also plans to build a number of AMSAT owned CubeSat Simulators and make them available to educators to use by shipping them around. To support this, we will document presentations and activities to be used with the CubeSat Sim. Ultimately, we hope to grow a community of satellite simulator enthusiasts who continue to improve and extend the CubeSat

Sim, adding additional functionality and satellite emulation modes.

## VII. ACKNOWLEDGMENTS

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## Authors' Note:

This article is the first article in a series that was published in the AMSAT Journal. The other parts are available in the AMSAT Journal or online at <http://cubsatsim.org/papers>